

**STUDY OF AERODYNAMICS ON THE ANALYSIS FOR
BLENDED WING BODY (BWB) UNMANNED AERIAL
VEHICLE (UAV) USING COMPUTATIONAL FLUID
DYNAMICS (CFD) FOR HIGH SUBSONIC FLOW**


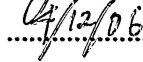
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A thesis submitted in partial fulfillment of the requirement for award of Bachelor
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DECEMBER 2006

“I declare that this thesis is the results of my own work except the ideas and summaries which I have clarified their sources. The thesis has not been accepted for any degree and is not concurrently submitted in candidature of any degree.”

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ACKNOWLEDGEMENT

First and foremost, Alhamdulillah and thanks to Allah the Almighty for the completion of this thesis and also am eternally grateful for the support and encouragement from my family especially my mother. Since the day I entered higher education institution, she has always inspired and motivates me beyond mere academic mediocrity. During the past year and the half, the support has continued as she has been a lonely hardworking and tough single mother; taking care of her own two children financially. The satisfaction of completing this project would have been meaningless without having my family support throughout the entire process. I dedicate this report for them.

I am also indebt to Mr. Rizal Effendy as my project advisor for his guidance into the finer aspects of making this thesis. The assistance of Mr. Aman as my co-advisor was invaluable as well not only trusting in accomplishing this project but also rather skilled CATIA guru. Moreover, this research would not have been possible without the help of Mr. Azli as a CFD genius which he gives insights into the finer aspects of CFD.

Last but not least, I would also like to forward my thanks to all my friends for all the help and support directly or indirectly for the completion of this thesis.

ABSTRACT

This thesis presents a study of Aerodynamic of a Blended Wing Body Unmanned Aircraft Vehicle (BWB UAV) within a UiTM project. A computational analysis is being carried out throughout the process through Computational Fluid Dynamics (CFD) software (FLUENT) which computed the steady, three-dimensional using the standard one-equation Spalart-Allmaras turbulence model. A comparison of meshing is being completed between the unrefined, refined, and adapted mesh sizes. The three-dimensional BWB Aerodynamic characteristics are validated using two-dimensional transonic airfoil at various angles of attack.

This thesis starts with literature overview then followed with mathematical model used in the analysis. Overview of Aerodynamics terminologies is presented in Chapter 3 and proceeds with the CFD general theory in the next chapter. Chapter 6 and 7 presents extensive discussions of numerical simulation and the outcome of simulation respectively throughout this project. This thesis is ended with conclusion and recommendation at the end of the chapter.

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